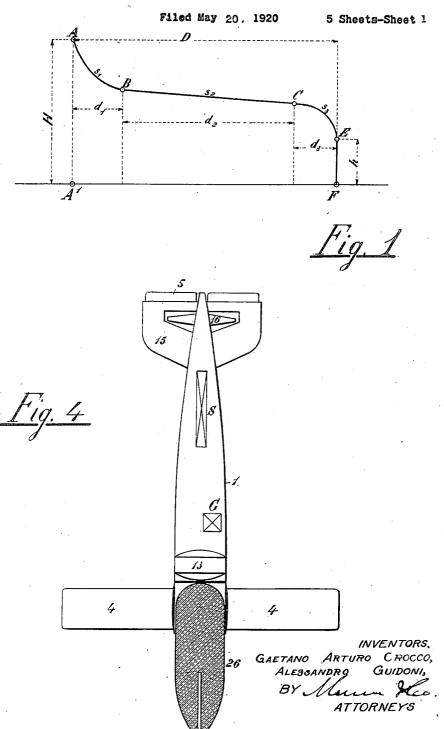
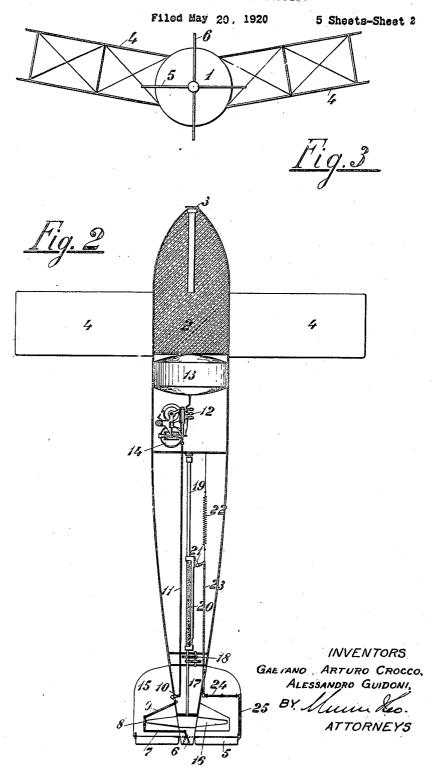
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## MOTORLESS AERIAL WINGED TORPEDO



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# MOTORLESS AERIAL WINGED TORPEDO

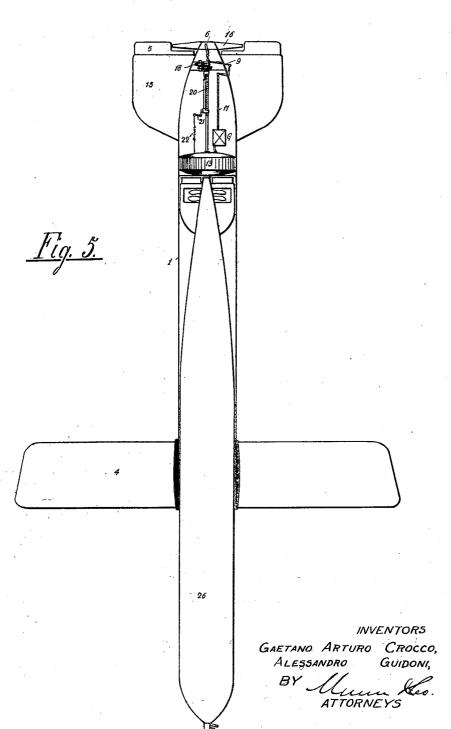


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MOTORLESS AERIAL WINGED TORPEDO

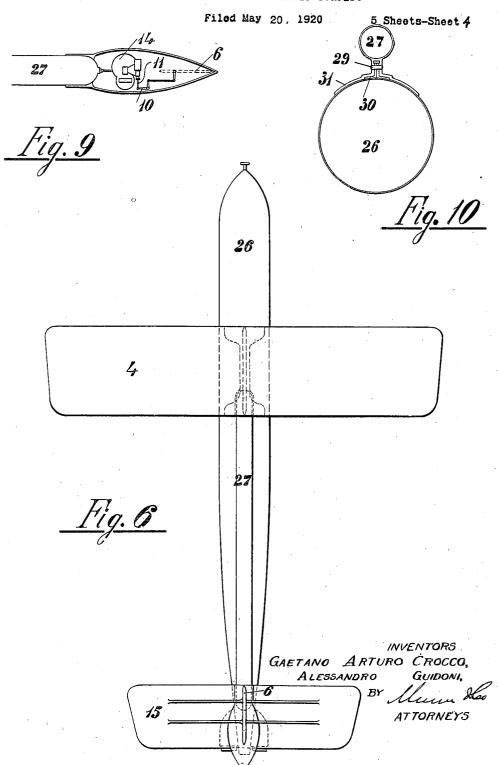
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G. A. CROCCO ET AL

#### MOTORLESS AERIAL WINGED TORPEDO

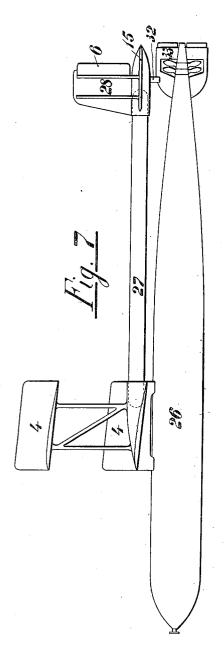


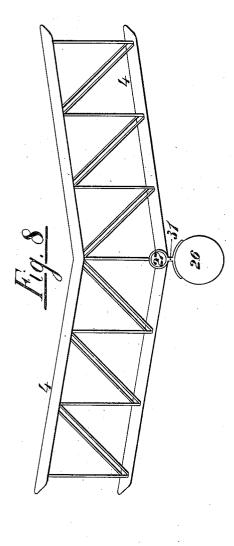
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MOTORLESS AERIAL WINGED TORPEDO

Filed May 20, 1920

5 Sheets-Sheet 5





INVENTORS

NO ARTURO CROCCO, ALESSANDRO GUIDONI, GAETANO

# UNITED STATES PATENT OFFICE.

GAETANO ARTURO CROCCO AND ALESSANDRO GUIDONI, OF ROME, ITALY.

MOTORLESS AERIAL WINGED TORPEDO.

Application filed May 20, 1920. Serial No. 382,932.

To all whom it may concern:

Be it known that we, GAETANO ARTURO Crocco, colonel of the Italian Army, and Alessandro Guidoni, engineer, both subjects 5 of the King of Italy, and residing at Rome, in the Kingdom of Italy, have invented certain new and useful Improvements in Motorless Aerial Winged Torpedoes, of which the following is a description.

Aerial bombardments must often be carried out against strongly defended places as cities or other important points provided with antiaerial defence, or against naturally strong places supplied with their own means 15 of defence as fortified positions, batteries,

warships and the like.

In such cases it is difficult for aircraft to approach the vertical of the target and remain there permanently in order to carry 20 out the bombardment and this must necessarily be done by keeping out of the vertical line of the target itself and possibly at a great distance therefrom.

The present invention refers to aerial <sup>25</sup> bombs and torpedoes that in cases as the aforesaid may be dropped from aircraft (airships and aeroplanes) upon targets that are out of their vertical line, even at a great

distance.

For such a purpose the bomb, shaped almost like a torpedo, or exactly like it when it is to be used against ships, and provided with steering surfaces which can be adjusted during its flight, is also supplied with supporting surfaces or wings shaped like those of an aeroplane allowing it to follow in a sliding flight a preestablished course up to a distance just enough to reach the target which is to be hit, or to descend to the surface of the water to lay down the torpedo which must afterwards follow its way until it reaches the ship to be torpedoed.

With the above mentioned objects in view the aerial torpedo, according to the present 45 invention, is provided with controlling means which during its flight ensure the steadiness of the course and limit the range of the flight as required; and, differently from other devices of the same kind, it does 50 net carry propelling machinery and is propelled solely by the action of its weight.

ally blowing and also for reducing the dimensions and the incumbrance of the appa- 55 ratus, to its bearing wings a very limited surface is assigned, so that to the gliding or sliding flight corresponds a very high speed, and, in order that this speed may be attained in a short time, the arrangement is such that 60 from the very beginning of its flight the torpedo falls almost vertically until it acquires the necessary speed while losing height, and, when the required speed has been reached, begin to enter into action certain devices 65 which, acting upon the steering surfaces,—depth rudders,—it will vary the position of them in such a way as to oblige the craft to proceed on its way in a sliding flight.

If there were not provided controlling 70 devices of the depth rudders and the latter were from the very beginning of the launching arranged in the position of the sliding flight, it is known that the apparatus would tend to acquire, in sliding downwards, an 75 excessive speed which would cause it to rise again afterwards and successively proceed according to a wavy trajectory of descending and ascending movements in wide sinu-

osities.

The action of the depth rudder set at the convenient angle and in due time, controls therefore the descent bringing in a ver; short time the apparatus in the proper position for the sliding flight.

The action of the device which control the rudders can be made dependent on any of the variable elements along the way: for instance on the time of the fall, on the length of the way covered, on the pressure of the 90 air, on the speed reached, or on the angle that the longitudinal axis of the torpedo forms the vertical line of descent.

In the form of execution of the torpedo which will be described by way of example, 95 there will be shown a device whose action is controlled by the length of the distance covered: it is understood, however, that any of the other methods mentioned above might just as well be conveniently adopted as they 100 still remain within the scope of the present invention.

By any of these means it is possible to cause the torpedo, abandoned to itself, to Besides, for the purpose of reducing to a start its flight by dipping down in rapid 10t minimum the drift due to the wind eventu- descent for a distance sufficient to attain

the necessary speed, a distance which is always the same whatever the distance from the target and the height of the aircraft,

from which it is dropped, may be.

In a second time the torpedo, righted by the depth rudders, covers a trajectory slightly inclined, in a steady sliding flight until it reaches the target; the length of this second distance depends on the distance of the target itself and must be regulated every time if not employing torpedoes destined to be dropped from a constant distance from the target.

At the end of this second stretch of its course the torpedo bends its trajectory downwards and descends vertically. In order to secure this result it is necessary to produce a considerable centripetal force by arranging the wings at an angle of negative incidence and this may be obtained by operating the depth rudder always set in the aforesaid manner in connection with one of the elements variable along the course.

It is understood that the result obtained by operating the rudders can be equally reached by varying the incidency of the Whichever the method followed be, when the trajectory of the bomb is thus reduced to be vertical, it needs a third action of the rudders in order to preserve the ob-

tained verticality of the falling line.

The operation is simplified when the target is a ship against which a torpedo must be launched, for in such a case the descent on the wings may be limited to its flight as far as to reach the surface of the sea where, as it will be explained hereunder, the torpedo abandons its winged support and begins its submerged course in the usual way towards the target.

In the drawing attached hereto illustrating schematically a form of execution of the aerial torpedo, according to the present

invention:

Fig. 1 shows a typical form of the trajectory

Fig. 2 is a horizontal axial section of the

torpedo:

Fig. 3 is a rear view in elevation of same; Fig. 4 shows a usual aerial torpedo carried by an independent winged support;

Fig. 5 shows a top view of a torpedo carried by an independent winged support which is seen in section for better showing the arrangement of the interior;

Figs. 6, 7 and 8 show a top view, a side and a front elevation respectively of a modification of the torpedo in an independent

winged support:

Fig. 9 is a detail of the control organs of the steering surfaces of the winged sup-

Fig. 10 is a detail of the attachment of the torpedo according to Figs. 6-8 to the winged support.

The aerial torpedo, Fig. 2, in its more general form, consists of a casing 1 shaped like a torpedo body and carrying in its head the explosive charge 2 with its fusee 3.

To the casing are attached the bearing 70 surfaces 4 which, to lessen the encumbrance, are biplane and arranged in the shape of an open V towards the upper part so as to concur in ensuring the lateral stability. On the other side both the size of the wings 75 and that of the steering surfaces, is rather limited owing to the high speed assigned to this torpedo in its sliding flight.

The steering surfaces are mounted on the tail of the torpedo-shaped body behind the 80 usual cross shaped fins and comprise a depth

rudder 5 and a steering rudder 6.

Through an articulated system of links and angle levers 7-8-9-10-11 the steering rudder 6 is connected to the rod of the 85 piston of a compressed air auxiliary motor 12 fed by a reservoir 13 and controlled by a spring gyroscope 14 of the same kind as those used for similar purposes in torpedoes so that it is not necessary to describe it herein more in detail.

Both of the two stabilizing fins 15 show little window to allow free passage to the blades of the distance measuring propeller 16, which, in the example illustrated 95 in the drawing constitutes the organ that, in connection with the distance covered,

must operate the depth rudder.

For this purpose the shaft 17, on which the propeller is keyed by means of a set of 100 gears designed to reduce the number of revolutions, transmits its movement to another shaft 19 bearing a screw-thread up to a certain portion of its length. Along this thread slides, during the revolving motion 105 of the propeller, a cam 20 conveniently shaped in accordance with the course that the torpedo must follow.

Against the cam rests, with an antifriction roller, one of the arms of an angle lever 21, the other arm of which is continually pulled by a spring 22 and at the other end connected, through an articulated system of links and levers 23—24—25 with

the depth rudder 5.

The working of the torpedo is as follows: through a convenient suspension system it is preferably carried at the lower side of the aircraft in a horizontal position, so that, during the flight, its weight is partially sup- 120 ported by the action of the wind which impinges against the wings 4.

The suspension system is preferably supported on a revolving pivot so as to allow that the dropping action may take place in 125 any direction whatsoever without having to change the course of the aircraft if the target should not be on its line of flight.

As soon as the aeroplane, or the airship, has reached the distance required from the

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target, the suspension device is released and it may be useful in practice to make use of while the torpedo gets free from its fastenings, the steering gyroscope whose spring had been wound up beforehand, begins to

In the first portion  $S_1$  of the trajectory, Fig. 1, extending from the starting point A whence the fall begins, up to point B at which the torpedo after having reached the required speed starts with the regular sliding flight the distance measuring propeller shifting the cam 20 operates the depth rudder in such a way as to cause the passage of the torpedo from its initial to its normal 15 flying position.

During the sliding flight, that is to say, within the distance  $s_2$  comprised between point B in which it starts and point C whence its downward movement begins, the steering cam keeps the rudders in the normal position, even while it continues to advance with a uniform motion along the screw-

threaded shaft.

Lastly, when the torpedo has reached point C, that is to say after having covered the distance  $s_1+s_2$ , corresponding to the horizontal course  $d_1+d_2$ , the cam operates the rudders so as to cause the bending of the trajectory downwards, i. e., along the course  $s_3$  which corresponds to the horizontal distance  $d_3$ . At the end of this bent portion, at point E, the torpedo and its rudders are in such a position that the wing action is stopped and the torpedo falls vertically upon the target F, without changing its course any more whatever the height —h— may be.

In order to regulate the course of the torpedo it will suffice to know the total horizontal distance  $D=d_1+d_2+d_3$  interceding between the vertical line AA' of the aircraft at the moment in which it is dropped and that of the target F. As the sum of the distances  $d_1+d_2$  is an experimental fixed quantity independent from the distance of the target and the height of the aircraft which drops the torpedo, the position of the cam is regulated according to the distance d<sub>2</sub> which is to be covered during the sliding flight.

This distance is assumed to be a preestablished and invariable one in the example given in the drawing, but it will be easily understood that the steering cam 20 may be shaped in such a way so as to enable to change at will the length of the distance

corresponding thereto.

distance measuring propeller may be caused to release, by one of the known ways, a safety device designed to prevent the premature explosion of the torpedo while it is

still near the aircraft.

In the above said it has been assumed, for greater generality, that the case referred to an aerial torpedo built completely anew. But and caudal steering surfaces.

the torpedoes already existing and destined

to be dropped vertically.

The result may be obtained by building separately the torpedo-shaped body with 70 the winged supporting surface, the stabilizing surfaces, and the steering organs, and by applying afterwards to this torpedo-shaped body, instead of charge 2 contained within—as shown in Fig. 2—a complete tor- 75 pedo 26 as shown in Fig. 4. The other organs, with which the torpedo-shaped body is provided, bearing surfaces 4, compressed air tank 13, gyroscope with its auxiliary motor and movable steering cam schemati- 80 cally shown by G and S, and the stabilizing device at the tail with the steering surfaces, remain unaltered. This arrangement, which is already very convenient for common bombs and torpedoes, for it allows to apply to the winged support torpedoes ready for use, and also of different patterns, eventually detachable from the winged support towards the end of their course, is most important in the case in which it is necessary 90 to drop the torpedo against a ship for it allows of using the torpedo, a very complicated device, just as it is without change in any of its parts.

The torpedo 26 is introduced into the tor- 95 pedo-shaped casing 1, formed of an external winged sheath—Fig. 5—in the same manner as in a torpedo launching tube, and it may be extracted therefrom at the moment in which the wing surfaces 4 meet with the water which, by its resistance stops them, in the same way as it issues from the tube above mentioned, when it is being launched, and it is in that moment therefore that enter into action, in the manner already 105 known, the gyroscope and the propelling apparatus which is to carry it upon the target, free from whatsoever appendages.

It is to be remarked moreover that in the case of a torpedo to be launched against a ship, the only parts which are needed of the whole trajectory shown in Fig. 1, are the portion  $s_1$  and, eventually, part of the sliding flight s2 if by change the apparatus were too high at the moment the torpedo is dropped. In this first part of the trajectory the steering organs connected with the winged casing 1, work in the identical manner of that explained before in describing the trajectory represented at Fig. 1.

A very convenient form of connection of In the same manner the movement of the the torpedo with the winged support is that shown in Figs. 6, 7 and 8. As it may be seen, the torpedo-shaped body 1 of the sheath is substituted with a tube 27 which constitutes in the same time the compressed air tank for the gyroscope auxiliary-motor, and the connection of the bearing biplane surface 4 with the stabilizing device 15—28

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These are simplified for, supposing practically to limit the speed of flight to values near that of the aircraft which drops the device, the depth rudder may be omitted as a separate element and formed into one piece together with the horizontal stabilizing device 15 conveniently sloping.

As a moveable piece there remains only the steering rudder 6 fastened to the verti-10 cal stabilizing surface 8 and therefore the movable steering cam 20, is done away with as well as the distance measuring propeller and there remains no other device but the steering gyroscope 14 which, with the com-15 pressed air auxiliary motor, controls in the usual manner the flight course rudder 6 dur-

ing the descent.

The suspension of the torpedo to the winged support is obtained by means of a 20 pin 29 passing through holes made in an extension or short dorsal fin 30 with which the torpedo is supplied, Fig. 10, and in the cheeks of a saddle-shaped seat or upset cradle 31 carried by the biplane cell. The 25 saddle-shaped seat is enough to keep in its place the torpedo, however, another seat 32 Fig. 7, placed under the end of tube 27 and resting on the vertical stabilizing fin 33 of the torpedo insures the rigidity of the connection.

In its fall the torpedo plunges first into the sea and then follows the winged support: the resistance of the pin 29 is regulated in such a way so as to ensure the break-35 ing through the resistance that the water makes against the wings, thus the torpedo gets free while the gyroscope and the propelling apparatus begin to work and runs towards the target leaving behind itself the winged support.

The disengaging devices necessary to initiate such a working have not been included

in the drawing for simplicity's sake.

1. An aerial torpedo to be dropped from aircraft upon targets lying distant from the vertical line of said aircraft, upon which targets it shall reach through the sole action of the gravity, comprising bearing surfaces sufficient to support it when in sliding flight, adjustable steering surfaces and means acting in correspondence to the distance traversed, and which control the position of said steering surfaces during the flight so as to steer the torpedo along a predetermined route in a vertical plane, the steering surfaces being successively set in such a manner that, as soon as the torpedo is left to itself,

it descends almost vertically in order to acquire a high speed with which to proceed 60 then in a sliding flight until it reaches the

vicinity of the target.

2. An aerial torpedo to be dropped from an aircraft upon a target lying far from the vertical line of said aircraft and com- 65 prising a body with supporting wings, a depth rudder provided on the body, a distance measuring propeller, a cam member, means for operating the cam from the measuring propeller, and means controlled by 70 the cam member for operating said rudder.

3. The combination of a winged support with stabilizing devices and steering surfaces, a tubular casing carried by said support, a complete self-moving water torpedo 75 with steering and propelling means contained in said casing, means as a distance measuring propeller acting on the steering means of the winged support to control its course while it moves downward under the 80 action of the gravity alone, and means to allow the torpedo leaving its casing and to start against a floating target, as soon as the whole device plunges into the water.

4. The combination of a winged support 85 with stabilizing devices and steering surfaces, a complete self-moving water torpedo with steering and propelling means freely suspended beneath said winged support, a breakable pin connecting the torpedo to the 90 winged support, means as a distance measuring propeller acting on the steering means of the winged support to control its course while, under the action of the gravity alone, it moves downward against a floating target, 95 the suspension pin breaking and releasing the torpedo when the wings come into contact with the water.

In testimony whereof we affix our signatures in the presence of two witnesses.

Signed by the first named applicant at Rome, in the Kingdom of Italy, this 31st day of March, 1920. GAETANO ARTURO CROCCO.

Witnesses for the first named applicant, Gaetano Arturo Crocco:

LETTERN LUBSCUTAS, Antonio Lubscutas.

Signed by the second named applicant at Washington, in the District of Columbia, this 17th day of May, 1920.
ALESSANDRO GUIDONI.

Witnesses for the second named applicant, Allesandro Guidoni:

GIUSEPPE A. F. SPARARRE. W. E. Beck, Jr.